

FINE & FAST STEERING MIRRORS

COMPACT - DYNAMIC - PRECISE

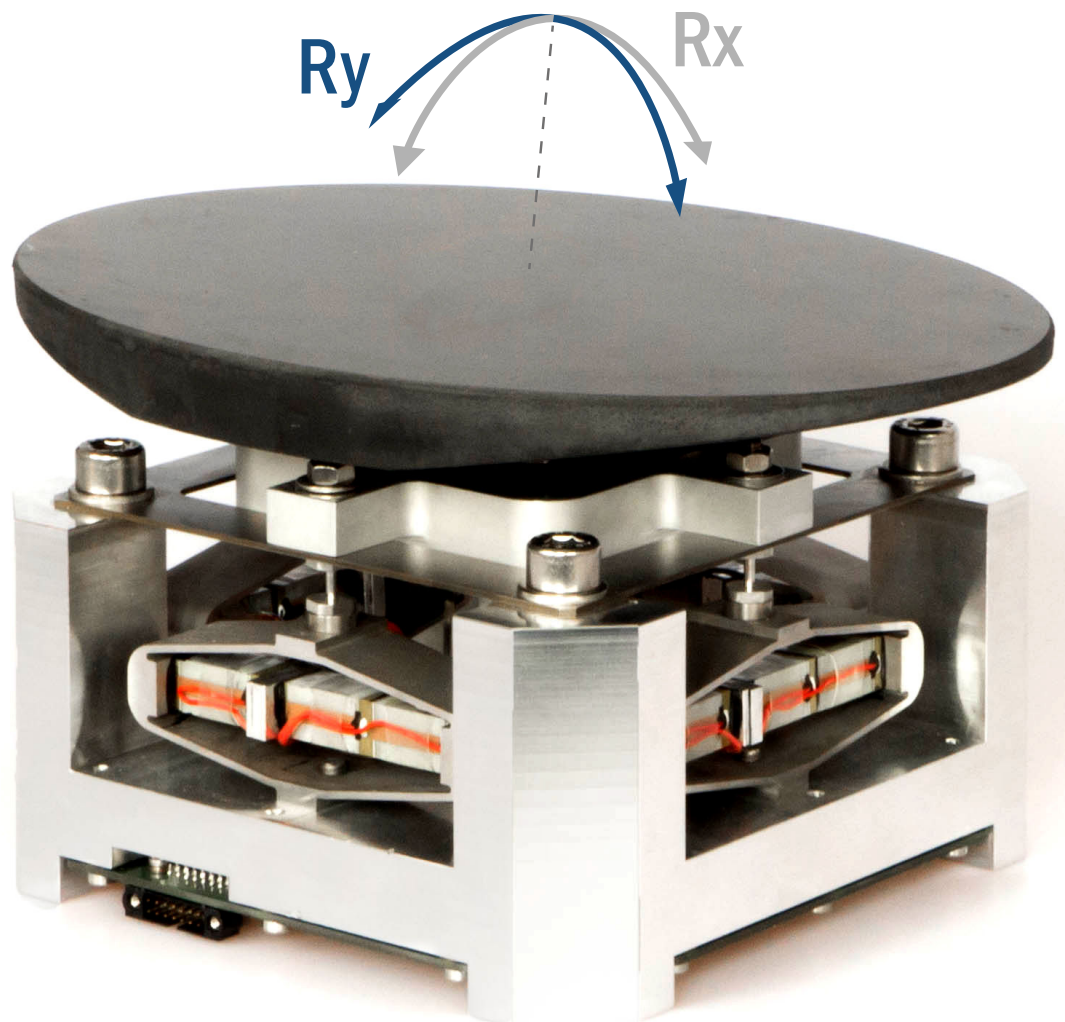


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(1) Peak-to-peak value. Mechanical angular stroke.

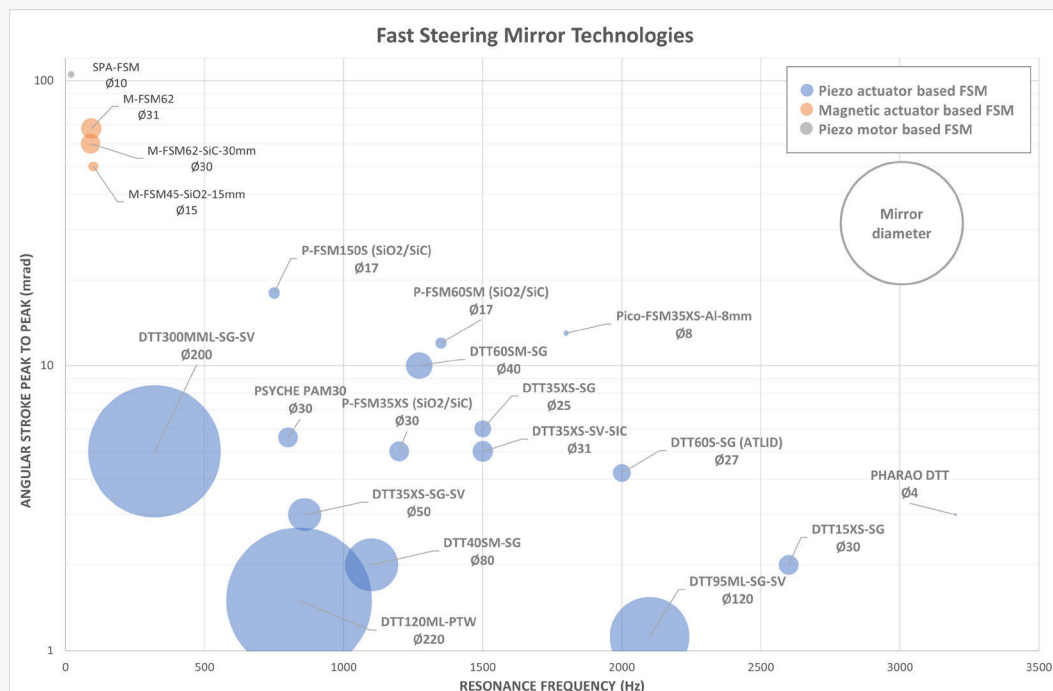
HERITAGE IN AEROSPACE & DEFENCE

CEDRAT TECHNOLOGIES (CTEC) piezo mechanisms and their associated electronics are widely used in Air, Space & Defence (ASD) applications. They can be found in optical instruments, cameras, telescopes and electro optic systems embedded on airplanes, helicopters, unmanned aerial vehicles (UAV), satellites, spacecrafts, etc... Their excellent dynamic performance, reliability and compactness make them ideal for the following mechatronic functions & applications:

- Image resolution enhancement (micro-scanning, pixel shift, dithering)
- Image & line of sight stabilisation
- Field of view increase
- Fillet compensation
- Fine pointing ahead
- Tracking

The DTT is the most common mechanism (steering platform) that constitutes a fast steering mirror (FSM), it means Double Tip Tilt i.e. tilt along two perpendicular axis. In the graph below, we show a list of FSM stroke vs resonance frequency vs mirror size.

In the previous table you will see a recap of the products realizing the steering function.



1. PHARAO DOUBLE TIP TILT (DTT)

1.1. DESCRIPTION

The Double Tip Tilt mechanism DTT35 XS-space delivered to SODERN (AIRBUS) is a very light piezoelectric mechanism (25 grams) designed according to space rules for the PHARAO ACES mission. The mechanism uses Strain Gauges (SG) as positioning sensor and allows to reach a 1:4000 stability (1 μ rad rms). It controls the incidence of a laser beam flux towards an optical fiber

1.2. APPLICATION

Space, fine pointing

1.3. ENVIRONMENTAL CONDITIONS

The DTT35 XS-space has followed a space qualification program according to ECSS standards (European Space Agency Standards)

- **Operating temperature in vacuum:** -20°C / +75°C
- **Random vibration:** 41 Grms
- **Lifetime:** 2 e8 cycles full stroke

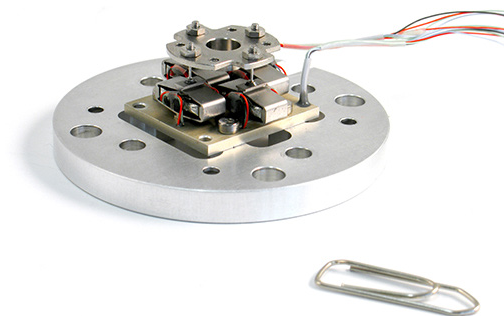


Fig. 1 : DTT35XS-space

PARAMETER	UNIT	PHARAO DOUBLE TIP TILT (DTT)
Max. angular stroke	mrad	3
Dimensions	mm	ø 30×22
Total mass	g	25
Mirror mass	g	1
Mirror dimensions	mm	ø 40 x 5
Loaded resonance frequency	Hz	3200
Position stability	μ rad	<1
Capacitance per axis	μ F	0.5

Table a : Performances of Pharao double tip tilt (DTT)

2. DTT60S-SG FOR ATLID

2.1. DESCRIPTION

Thanks to the heritage from the PHARAO DTT, CTEC has developed the ATLID Beam Steering Mechanism (BSM) for SODERN (see publication). ATLID is a Lidar instrument for the EarthCARE mission. The BSM is a Tip-Tilt piezo mechanism based on 4 APAs including Strain Gauges. The requirements were particularly severe regarding the long term stability and the cleanliness. CTEC has successfully delivered the Flight Models in 2015

The BSA (Beam Steering Assembly), included in the emission path, aims at deviating a pulsed high energy UV laser beam to compensate the pointing misalignment between the emission and reception paths of ATLID. It requires a very high stability and high resolution

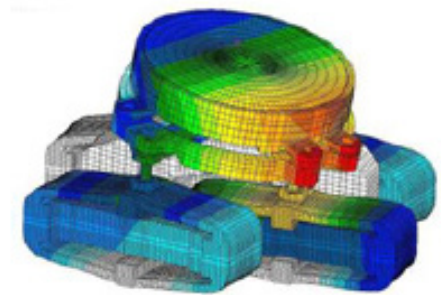


Fig. 2 : BSM CAD model

2.2. APPLICATION

Space

2.3. ENVIRONMENTAL CONDITIONS

- Non magnetic
- **Shock level:** 100 G
- **Random Vibration level:** 15.5 Grms
- **Quasi Static:** 26 g
- **Cleanliness:**
 - **Particular:** 50 ppm
 - **Molecular:** 5.10 e-8 g/cm²

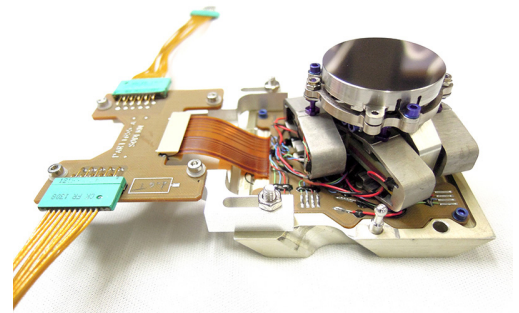


Fig. 3 : Complete BSM mechanism

PARAMETER	UNIT	DTT60S-SG FOR ATLID
Max. angular stroke Rx	mrاد	4.2
Max. angular stroke Ry	mrاد	3
Dimensions	mm	63×62×40
Total mass	g	130
Mirror mass	g	2.9
Mirror dimensions	mm	∅ 27×5
Loaded resonance frequency	Hz	2000
Resolution	nrad	400
Repeatability	μrad	70
Capacitance per axis	μF	3

Table b : Performances of DTT60S-SG for ATLID

3. POINT AHEAD MECHANISM (PAM30)

3.1. DESCRIPTION

The Point Ahead Mechanism (PAM30) consists in a SiC mirror mounted on a 2-rotation axis ("Tip Tilt"), small range and pointing mechanism with Strain Gages positioning sensors, mounted on a bracket including a thermal sensor.

3.2. APPLICATION

The PAM30 aims at accurately pointing the optical downlink signal (communication laser beam) towards Earth position from the Psyche Asteroid beyond Mars Planet.

The Psyche mission, led by NASA JPL, will be the opportunity to test the first deep-space optical communication made from a satellite to earth.

3.3. ENVIRONMENTAL CONDITIONS

- **Operating temperature:** -40/+65 °C
- **Vacuum compatible**
- **Random vibration:** up to 42 Grms
- **Lifetime:** > 7 years

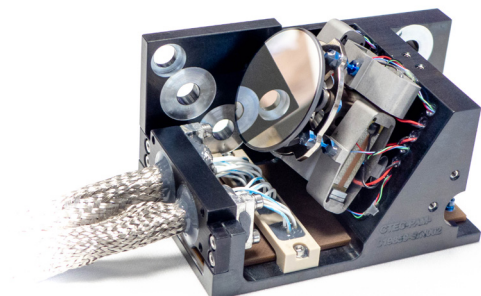


Fig. 4 : PAM30

PARAMETER	UNIT	PAM30
Max. angular stroke	mrad	8
Total mass	g	< 500
Mirror dimensions	mm	ø 32
Loaded resonance frequency	Hz	> 1200
Reflective Wavefront Error PV	nm	40

Table c : Performances of PAM30

4. DTT15XS-SG

4.1. DESCRIPTION

Compact FSM with its controller box for embedded electro optic systems

4.2. APPLICATION

Image stabilisation and micro-scan combined in a single unit (5th and 6th axis of gyro stabilised platform).

4.3. ENVIRONMENTAL CONDITIONS

- **Operating temperature:** -40°C/+70°C
- **Storage temperature:** -55°C/+85°C
- **High altitude operation:** 55 000 ft
- **High altitude storage:** 70 000 ft
- **Half sine Shock level :** >20 G
- **Vibration level (CBO):** 0.025 g²/Hz up to 1 kHz

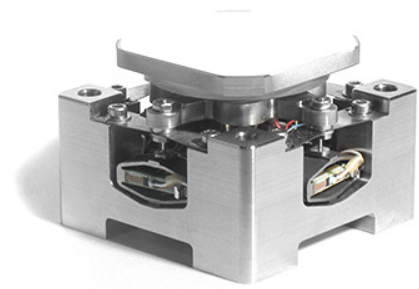


Fig. 7 : DTT15XS-SG



Fig. 5 : Batch of DTT15XS-SG with mirror



Fig. 6 : DTT15XS-SG with CCBu20 controller

PARAMETER	UNIT	DTT15XS-SG
Max. angular stroke	mrad	2
Dimensions	mm	40×40×24
Total mass	g	< 110
Mirror mass	g	10
Mirror dimensions	mm	30×30×4.85
Loaded resonance frequency	Hz	2600
Resolution	nrad	50
Settling time within 5%	ms	1
Position stability	μrad	<25
Capacitance per axis	μF	0.6
Power consumption with CCBu20 50 Hz @ +/- 700 μrad	W	9
Power consumption with CCBu20 480 Hz @ +/- 200 μrad	W	13.5
Controller	Driven by custom embedded CTEC electronic and controller board call CCBu20-PROX	
Embedded electronics board	SG conditioning, EEPROM memory, temperature monitoring and compensation	

Table d : Performances of DTT15XS-SG

5. DTT35XS-SG-SV

5.1. DESCRIPTION

FSM for 50 mm diameter mirror with CCBu20 controller box

5.2. APPLICATION

Line of sight stabilisation

5.3. ENVIRONMENTAL CONDITIONS

- **Operating temperature range:** -40°C to +70°C
- **Shocks :** 30 G



Fig. 8 : DTT35XS-SG-SV with CCBu20 electronics

PARAMETER	UNIT	DTT35XS-SG-SV
Max. angular stroke	mrاد	3
Dimensions	mm	ø 48×27
Total mass	g	164
Mirror mass	g	25
Mirror dimensions	mm	ø 50×6
Loaded resonance frequency	Hz	860
Resolution	μrad	< 1
Settling time within 5%	ms	<2
Linearity	%	0.2
Control bandwidth	Hz	200
Capacitance per axis	μF	0.5
Controller	CTEC standard CCBu20	
Embedded electronics board	SG conditioning, EEPROM memory, temperature monitoring and compensation	

Table e : Performances of DTT35XS-SG-SV

6. DTT35XS-SV-SiC

6.1. DESCRIPTION

FSM for 31 mm diameter mirror with CCBu20 controller box

6.2. APPLICATION

Line of sight stabilisation, Laser Pointing Ahead

This version of DTT35XS is especially designed for vacuum space application

6.3. ENVIRONMENTAL CONDITIONS

- **Operating temperature range:** -40°C to +70°C
- **Shocks :** 30 G
- **Vacuum compatible**



Fig. 9 : DTT35XS-SV-SiC

PARAMETER	UNIT	DTT35XS-SV-SiC
Max. angular stroke	mrad	5
Dimensions	mm	ø 45×22
Total mass	g	65
Mirror mass	g	5
Mirror dimensions	mm	ø 31×6
Loaded resonance frequency	Hz	2400
Resolution	μrad	< 1
Settling time within 5%	ms	5 at full stroke
Linearity	%	0.2
Control bandwidth	Hz	>600
Capacitance per axis	μF	0.5
Controller	CTEC standard CCBu20	

Table f : Performances of DTT35XS-SV-SiC

7. DTT40SM-SG

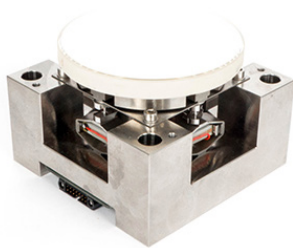


Fig. 10 : DTT40SM-SG

7.1. DESCRIPTION

FSM for 80 mm diameter mirror

7.2. APPLICATION

High power laser, atmospheric disturbance compensation

7.3. ENVIRONMENTAL CONDITIONS

- CTEC commits on mirror flatness in operations
- **Operation temperature:** -10/+50 °C
- **Storage temperature:** -20/+70 °C
- **Vibrations:** random 4 Grms up to 500 Hz
- **Shocks:** 20 G/11 ms half sine

PARAMETER	UNIT	DTT40SM-SG
Max. angular stroke	mrاد	2
Dimensions	mm	110×110×63
Total mass	g	1600
Mirror mass	g	100
Mirror dimensions	mm	ø80×8
Loaded resonance frequency	Hz	1100
Settling time within 5%	ms	2.6
Accuracy	μrad	<1
Linearity	%	<0.25
Rising time 90%	μs	780
Capacitance per axis	μF	3
INVAR components to reduce the mechanism CTE and mismatch with the optical component		

Table g : Performances of DTT40SM-SG

8. DTT60SM-SG

8.1. DESCRIPTION

The DTT60SM-SG is a piezo Tip Tilt Platform for Fast Steering Mirror (FSM) for electro-optics systems

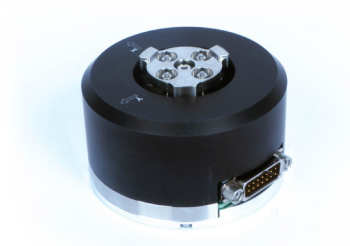


Fig. 11 : DTT60SM-SG

8.2. APPLICATION

Line of sight stabilisation inside electro-optic systems

8.3. ENVIRONMENTAL CONDITIONS

- **Operation temperature:** $-40^{\circ}\text{C}/+70^{\circ}\text{C}$



Fig. 12 : Rack RK42F3U

PARAMETER	UNIT	DTT60SM-SG
Max. angular stroke	mrad	10
Dimensions	mm	ø 65x40
Total mass	g	310
Mirror mass	g	16
Mirror dimensions	mm	ø 40x5
Unloaded resonance frequency	Hz	1746
Loaded resonance frequency (Blocked- free with 26mm x 6mm BK7 mirror)	Hz	1 272
Loaded resonance frequency (Blocked- free with 40mm x 5mm BK7 mirror)	Hz	806
Capacitance per axis	μF	3
Resolution	μrad	1
Settling time	ms	2
Controller	LA75B amplifier into a desktop rack RK42F-3U	

Table h : Performances of DTT60SM-SG

9. DTT10M-SG-SV

9.1. DESCRIPTION

FSM for 50 mm diameter mirror with its controller rack

9.2. APPLICATION

High power Laser, atmospheric disturbance compensation

9.3. ENVIRONMENTAL CONDITIONS

- CTEC commits on mirror flatness in operations
- **Operation temperature:** -10°C/+50°C
- **Storage temperature:** -20°C/+70°C
- **Vibrations:** Random 4 Grms up to 500 Hz
- **Shock:** Half Sine 20 G 11 ms



Fig. 13 : DTT10M-SG-SV and electronic rack

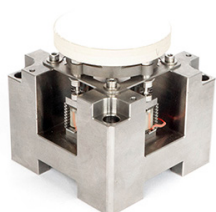


Fig. 14 : DTT10M-SG-SV

PARAMETER	UNIT	DTT10M-SG-SV
Max. angular stroke	mrad	0.5
Dimensions	mm	65×65×65
Total mass	g	860
Mirror mass	g	30.25
Mirror dimensions	mm	ø50×7
Mirror flatness		L/10 (632 nm) over CA 90 %
Loaded resonance frequency	Hz	5100
Settling time within 5%	ms	0.7
Accuracy	μrad	<1
Linearity	%	<0.25
Control bandwidth	Hz	1000
Capacitance per axis	μF	1
INVAR material to reduce the mechanism CTE and mismatch with the optical components		

Table i : Performances of DTT10M-SG-SV

10. DTT95ML-SG-SV

10.1. DESCRIPTION

FSM with 120 mm diameter SiC mirror

10.2. APPLICATION

Space, point-ahead mechanism for infrared telescope

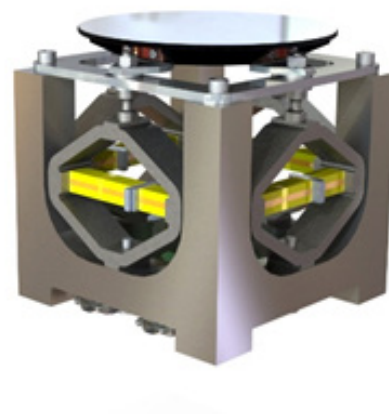


Fig. 15 : CAD view of DTT95ML-SG-SV

10.3. ENVIRONMENTAL CONDITIONS

- SRS 800 G
- Random 8 Grms

PARAMETER	UNIT	DTT95ML-SG-SV
Max. angular stroke	mrad	1.12
Dimensions	mm	127×127×128
Mirror dimensions	mm	>130
Loaded resonance frequency	Hz	2100
Control bandwidth	Hz	700
Capacitance per axis	μF	40
Embedded electronics board	Embedded electronics board to secure the wires	
	Driven with 0-80V reduced voltage	

Table j : Performances of DTT95ML-SG-SV

11. DTT120ML-PTW FOR MEFISTO

11.1. DESCRIPTION

In the context of CNES future observation satellite and DGA funding, and in collaboration with CNES and SODERN, CTEC designed a large space mechanism that allows to tilt a 2 kg payload by 0.5 mrad in 2 ms with a tracking error of less than 1%, while the Loaded resonance frequency is in the 500 Hz range. The vibrations created by the payload are compensated by moving another mass in the opposite direction. It uses 8 APA120 ML with strain gauges

This project allowed CTEC to create software to compensate the vibrations with reduced additional masses

11.2. APPLICATION

The MEFISTO mechanism is dedicated to fillet compensation for space telescopes

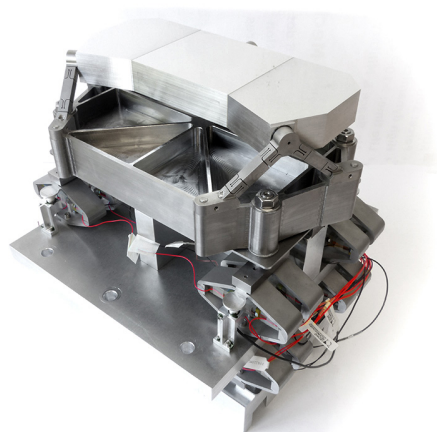


Fig. 16 : View of MEFISTO mechanism

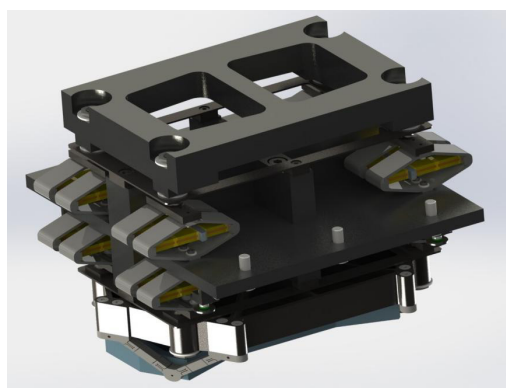


Fig. 17 : CAD of MEFISTO mechanism

PARAMETER	UNIT	DTT120ML-PTW FOR MEFISTO
Max. angular stroke	mrad	1.5
Dimensions	mm	279×250×293
Total mass	g	12 700
Mirror mass	g	730
Mirror dimensions	mm	220×96×23
Loaded resonance frequency	Hz	840
Speed		0.5 mrad in 2 ms
Maximum error	μm	2
Capacitance per axis	μF	40
		Capacitive sensors

Table k : Performances of DTT120ML-PTW for Mefisto

12. DTT300ML-SG-SV

12.1. DESCRIPTION

FSM for SiC large aperture mirror with its controller rack



Fig. 18 : FSM with non-coated SiC mirror

12.2. APPLICATION

High power laser, line of sight stabilisation for atmospheric disturbance compensation

12.3. ENVIRONMENTAL CONDITIONS

- -20 °C to +55 °C



Fig. 19 : Customised controller

PARAMETER	UNIT	DTT300ML-SG-SV
Max. angular stroke	mrad	5
Dimensions	mm	145×145×100
Total mass	g	1800
Mirror mass	g	900
Mirror dimensions	mm	200×140×36
Loaded resonance frequency	Hz	320
Resolution	nrad	200
Accuracy	μrad	3
Accuracy	%	2
Control bandwidth	Hz	>100
Capacitance per axis	μF	40
Embedded electronics board	Embedded electronics board for Strain Gauges conditioning	

Table I : Performances of DTT300ML-SG-SV

13. CRYOGENIC DTT10H FOR ESA CFSM PROJECT

13.1. DESCRIPTION

For the ESA EChO mission, the use of a fine steering tip and tilt mechanism ensures the stability of the line of sight for a telescope operating in cryogenic conditions. The main efforts were focused on the management of thermo-mechanical behaviour and performances of the actuator over a wide temperature range. Its capabilities to sustain stresses due to thermal cycles between room and cryogenic temperatures have been demonstrated through the development and test of an Engineering Model

13.2. APPLICATION

Space, pointing

13.3. ENVIRONMENTAL CONDITIONS

- Cryogenic stability 30 K
- Non-magnetic mechanism
- **Successful life time @30 K:** 2.2 M cycles
- **Successful vibration tests:**
 - **Shocks:** 700 G
 - **Random:** 18 Grms

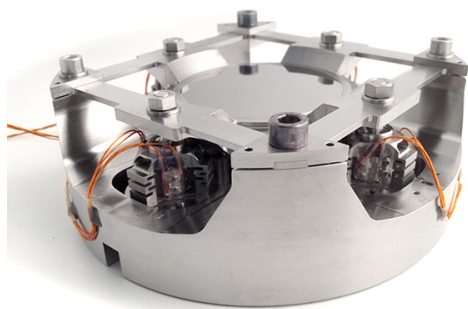


Fig. 20 : DTT10H cryogenic mechanism

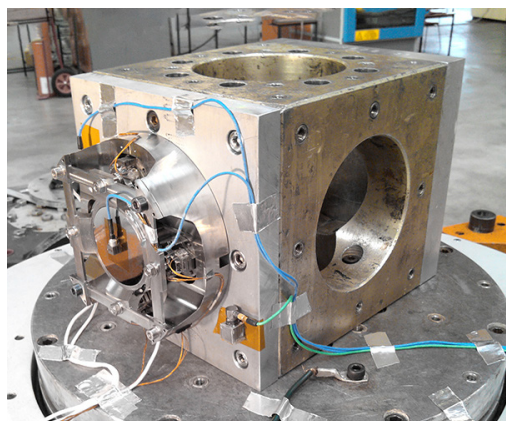


Fig. 21 : Shocks and vibrations test bench

PARAMETER	UNIT	CRYOGENIC DTT10H FOR ESA CFSM PROJECT
Max. angular stroke @300 K	mRad	0.2
Max. angular stroke @30 K	mRad	0.06
Dimensions	mm	ø 145×55
Total mass	g	1650
Mirror mass	g	72
Mirror dimensions	mm	ø 60×6
Loaded resonance frequency	Hz	3300
Resolution	nrad	0.13
Power consumption	mWatt	2
Piston drift along qualification	µm	<50

Table m : Performances of cryogenic DTT10H for ESA CFSM project

14. MAGNETIC FAST STEERING MIRROR (M-FSM62)

14.1. DESCRIPTION

Magnetic FSM for optical pointing with larger motion than piezo FSM while keeping high resolution and large bandwidth.

The M-FSM62 mechanism volume contains the mirror, the actuators, the flexure bearings and the Eddy Current positioning Sensors (ECP).

High performance Moving Iron Control Actuator (MICA™) allows to perform the dynamic strokes with low Joule heating.

Frictionless flexure bearings allow to achieve both high resolution and infinite life time.

Dedicated electronics for sensing, driving and controlling are respectively the [ECS45](#) and [MCSA480](#).



Fig. 22 : Magnetic Fast Steering Mirror
M-FSM62 (with Ø31 mm mirror)

14.2. APPLICATION

Typical applications are Free Space Optic (FSO) communication links, applications, optical imaging.

PARAMETER	UNIT	MAGNETIC FAST STEERING MIRROR (M-FSM62)
Max. angular stroke	mrad	140
Dimensions	mm	Ø 62 × 60
Standard SiC mirror size	mm	Ø 31
Mirror full stroke open loop bandwidth	Hz	200
(with heating)	Hz	93
Resolution	μrad	20
Loaded resonance frequency	Hz	100
Power @20Hz	W	0.5
See more detailed performances		

Table n : Performances of Magnetic Fast Steering Mirror (M-FSM62)

15. M-FSM45

15.1. DESCRIPTION



Fig. 23 : M-FSM45

The MFSM45 is a Magnetic based Fast Steering Mirror of +15 mm diameter clear aperture (in SiC, metallic or glass substrate) with 100 mrad angular stroke (peak-to-peak). The MFSM45 can be equipped with Eddy Current positioning sensor (ECS) for accurate positioning control.

15.2. APPLICATION

Fast steering mirror, Point ahead mechanism, Line of sight stabilisation, Microscanning, Tracking, Fine pointing.

PARAMETER	UNIT	MAGNETIC FAST STEERING MIRROR (M-FSM45)
Max. angular stroke	mrad	50
Dimensions	mm	Ø45 × 40
Standard SiC mirror size	mm	Ø17
Mirror full stroke open loop bandwidth	Hz	350
Resolution	μrad	<0.5
Loaded resonance frequency	Hz	100
See more detailed performances		

Table o : Performances of Magnetic Fast Steering Mirror (M-FSM45)

16. PIEZO FAST STEERING MIRROR (P-FSM150S-SG)

16.1. DESCRIPTION

The P-FSM150S is a piezo based Fast Steering Mirror of +15 mm diameter clear aperture (in SiC, metallic or glass substrate) with +18 mrad angular stroke. The P-FSM150S is equipped with Strain Gages positioning sensor for accurate positioning control. The P-FSM150S has been qualified for Space environmental conditions.

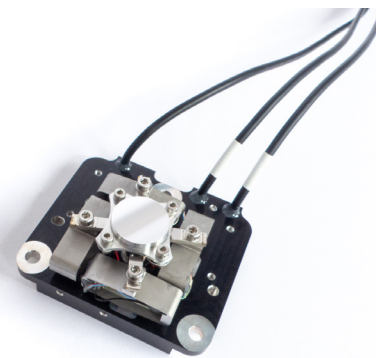


Fig. 24 : P-FSM150S-SG

16.2. APPLICATION

Fast steering mirror, Point ahead mechanism, Line of sight stabilisation, Microscanning, Tracking, Fine pointing.

PARAMETER	UNIT	PIEZO FAST STEERING MIRROR (P-FSM150S)
Max. angular stroke	mrad	18
Dimensions	mm	61 x 63 x 30
Standard SiC mirror size	mm	Ø17
Mirror full stroke open loop bandwidth	Hz	400
Resolution	µrad	<2
Loaded resonance frequency	Hz	700
See more detailed performances		

Table p : Performances of Magnetic Fast Steering Mirror (P-FSM150S)

17. MINI FAST STEERING MIRROR (MINI P-FSM35XS)

17.1. DESCRIPTION

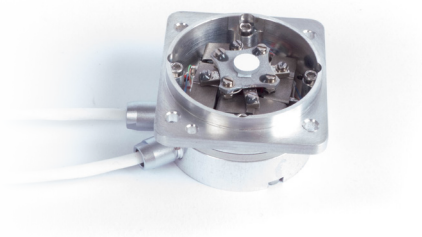


Fig. 25: View of the Mini P-FSM35XS

Recently, CTEC has delivered a mini FSM for a 3U CubeSat, for an undisclosed constellation that will have several hundreds of satellites. This mini FSM offers a **stroke of 6 mrad and a resonant frequency of 1700 Hz, with a mass of 50 gr.**

This FSM mechanism is a good candidate for all projects involving nano satellites or cube satellites, and is a high end alternative to MEMS, which reliability and resistance to space conditions are still being questioned.

To control the Mini FSM, CTEC developed a custom 2-channel CCBu20 controller qualified for vibrations. The controller prototype will be used for a New Space mission and needs to survive the extreme vibration conditions met during launch. To ensure the resistance of the controller, CTEC designed a new reinforced housing and performed vibration tests.



Fig. 26 : View of the CCBu20-SV

17.2. APPLICATIONS

Fast steering mirror, point ahead mechanism, line of sight stabilisation, microscanning, tracking, fine pointing

PARAMETER	UNIT	MINI P-FSM35XS
Max. angular stroke	mrad	12
Resonant frequency with 1x5mm mirror (Blocked Free)	Hz	1700
Angular resolution	μrad	2.2
Operating temperature	°C	-55 to 80
Height without mirror	mm	20
Typical mass	g	50

Table q : Mini P-FSM35XS Specifications

CEDRAT TECHNOLOGIES (CTEC) offers off-the-shelf mechatronics products including piezoelectric & magnetic actuators, motors, mechanisms, transducers and sensors with corresponding drivers & controllers. These mechatronics products are used for scientific and industrial applications requiring fonctions such as: micro and nano positioning, generation of vibrations, micro-scanning, fast & precise motion control, active control of vibrations, and energy harvesting

Most of the products are available in OEM versions for low cost and high volume industrial applications. CTEC also offers services including, design, R&D under contract and training

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CTEC is a SME located in Meylan, Inovallée, the French Innovation Valley near Grenoble. CTEC is recognised as a highly innovative company and has received several awards

CEDRAT TECHNOLOGIES

59 Chemin du Vieux Chêne - Inovallée
38246 Meylan Cedex

+33 (0)4 56 58 04 00
www.cedrat-technologies.com

